



**K.S. SCHOOL OF ENGINEERING AND MANAGEMENT, BANGALORE - 560109**  
**DEPARTMENT OF MECHANICAL ENGINEERING**  
**I SESSIONAL TEST QUESTION PAPER 2018 – 19 ODD SEMESTER**  
**SET-A**

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**Degree** : B.E  
**Branch** : Mechanical Engineering  
**Course Title** : Turbo Machines  
**Duration** : 90 Minutes

**Semester** : V  
**Date** : 4-9-2019  
**Course Code** : 17ME53  
**Max Marks** : 30

**Note: Answer ONE full question from each part**

Q. No.	Question	Marks	K Level	CO mapping
<b>PART-A</b>				
1(a)	With suitable velocity triangles, <b>derive</b> an expression for maximum hydraulic efficiency of a Pelton wheel in terms of blade velocity coefficient and blade discharge angle.	5	K3 Applying	CO1
(b)	A 137 mm diameter jet of water issuing from a nozzle impinges on the buckets of a Pelton wheel and the jet is deflected through an angle of $165^\circ$ by the buckets. The head available at the nozzle is 400m. Assuming coefficient of velocity as 0.97, speed ratio as 0.46 and reduction in the relative velocity while passing through the buckets as 15%, <b>find:</b> (i) Force exerted by the jet on the buckets in the tangential direction (ii) the power developed.	5	K3 Applying	CO1
(c)	<b>Define</b> a Turbo machine. With a neat sketch <b>explain</b> the parts of a turbo machine.	5	K2 Understanding	CO2
<b>OR</b>				
2(a)	<b>Define</b> and <b>explain</b> the following efficiencies of a hydraulic turbine: (i) Hydraulic efficiency (ii) Mechanical Efficiency (iii) Overall efficiency	5	K2 Understanding	CO1
(b)	A double jet Pelton wheel is required to generate 7500kW when the available head at the base of the nozzle is 400m. The jet is deflected through $165^\circ$ and the relative velocity of the jet is reduced by 15% in passing over the buckets. <b>Determine:</b> (i) diameter of each jet (ii) Total flow rate (iii) Force exerted by the jet in the tangential direction. Assume Overall efficiency = 80%, Speed ratio = 0.47, $C_v = 0.97$ .	5	K3 Applying	CO1
(c)	<b>Explain</b> how turbo machines are classified.	5	K2 Understanding	CO2

<b>PART-B</b>				
<b>3(a)</b>	<b>Draw</b> a neat sketch of a Francis turbine and <b>explain</b> the functions of each part. <b>Draw</b> the velocity triangles of a Francis turbine.	<b>5</b>	<b>K2</b> Understanding	<b>CO1</b>
<b>(b)</b>	The following data is given for a Francis turbine: Net head = 70m, Speed = 600RPM, Shaft power = 368kW, Overall efficiency = 85%, Hydraulic efficiency = 95%, Flow ratio = 0.25, Breadth ratio = 0.1, Outer diameter of the runner = 2 x inner diameter of the runner. Velocity of flow is constant at outlet and inlet. The thickness of the vanes occupies 10% of the circumferential area of the runner and discharge is radial at outlet. <b>Determine:</b> (i) Guide blade angle (ii) Runner vane angles at inlet and outlet (iii) Diameter of runner at inlet and outlet (iv) width of the runner at inlet.	<b>5</b>	<b>K3</b> Applying	<b>CO1</b>
<b>(c)</b>	<b>Differentiate</b> between a turbo machine and a positive displacement machine.	<b>5</b>	<b>K2</b> Understanding	<b>CO2</b>
<b>OR</b>				
<b>4(a)</b>	<b>Draw</b> a neat sketch of a Kaplan turbine and <b>explain</b> the functions of each part. <b>Draw</b> the velocity triangles of a Kaplan turbine.	<b>5</b>	<b>K2</b> Understanding	<b>CO1</b>
<b>(b)</b>	A Kaplan turbine working under a head of 20m develops 11775kW. The outer diameter of the runner is 3.5m, hub diameter is 1.75m. The guide blade angle at the extreme edge of the runner is 35°. The hydraulic and overall efficiencies of the turbine are 0.88 and 0.84 respectively. If the velocity of whirl at outlet is zero, <b>determine:</b> (i) runner vane angles at inlet and outlet at the extreme edge of the runner (ii) speed of the turbine.	<b>5</b>	<b>K3</b> Applying	<b>CO1</b>
<b>(c)</b>	<b>Classify</b> the following as Power generating or Power absorbing turbo machine: (i) Steam Turbine (ii) Fan (iii) Air blower (iv) Axial flow compressor (v) Wind Turbine	<b>5</b>	<b>K2</b> Understanding	<b>CO2</b>

Course In charge

Head - Dept

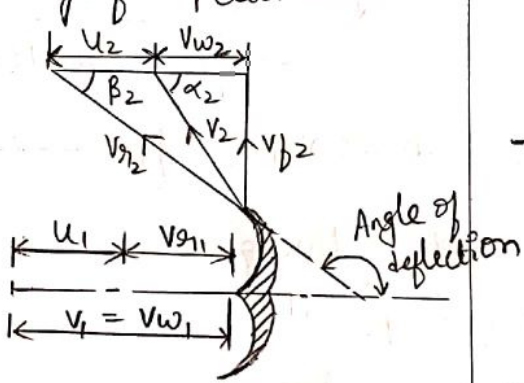
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Q. No.	Questions with Scheme & Solution	Marks	K Level	CO mapping
<b>PART-A</b>				
1(a)	<p>Expression for maximum efficiency of a Pelton wheel.</p> $V_1 = C_v \cdot \sqrt{2gh}$ $u_1 = u_2 = \frac{\pi D N}{60}$ $v_{w1} = v_1$ <p>Work done = <math>F_x \times u</math> per second</p> $W \cdot D / s = \rho a v_1 \times u [v_{w1} + v_{w2}] \quad \text{--- (1)}$ $\eta_H = \frac{\text{Work done per second}}{\text{Kinetic Energy of jet}} = \frac{\rho a v_1 u [v_{w1} + v_{w2}]}{\frac{1}{2} \rho a v_1 \cdot v_1^2}$ $\eta_H = \frac{2u [v_{w1} + v_{w2}]}{v_1^2} \quad \text{--- (2)}$ <p>Efficiency will be maximum when <math>\frac{d}{du} (\eta_H) = 0</math></p> $\Rightarrow u = \frac{v_1}{2} \quad \text{--- (3)} \quad \text{Substituting (3) in (2)}$ $\therefore \eta_{Hmax} = \frac{(1 + \cos \beta_2)}{2}$	5	K3	CO1
		1-		
		1-		
		1-		
		1-		

1.(b)

5

K3

Given: diameter of jet,  $d = 0.137 \text{ m}$  Velocity triangle

$$\text{area of jet, } a = \frac{\pi}{4} \times 0.137^2 = 0.014 \text{ m}^2$$

$$\text{Angle of deflection, } (180 - \beta_2) = 165^\circ \Rightarrow \boxed{\beta_2 = 15^\circ}$$

$$\text{Net Head, } H = 400 \text{ m}$$

$$C_v = 0.97, \text{ Speed ratio} = 0.46$$

$$\text{Relative velocity at outlet} = (1 - 0.15) \times \text{relative velocity at inlet.}$$

$$\therefore \boxed{V_{r_2} = 0.85 V_{r_1}}$$

$$V_1 = C_v \sqrt{2gH} \Rightarrow \boxed{85.93 \text{ m/s} = V_1} \Rightarrow \boxed{V_{w_1} = 85.93 \text{ m/s}}$$

$$u_1 = \phi \sqrt{2gH} \Rightarrow \boxed{u_1 = 40.75 \text{ m/s}} = u_2 = u$$

$\therefore$  From inlet velocity triangle,  $V_{r_1} = V_1 - u$

$$\Rightarrow \boxed{V_{r_1} = 45.18 \text{ m/s}}$$

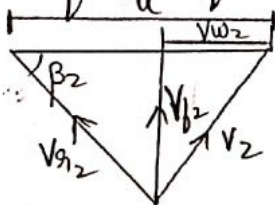
Sol W.K.T.  $V_{r_2} = 0.85 V_{r_1} \Rightarrow \boxed{V_{r_2} = 38.4 \text{ m/s}}$

From outlet velocity triangle,

$$V_{w_2} = V_{r_2} \cos \beta_2 - u$$

$$\boxed{V_{w_2} = -3.65 \text{ m/s}}$$

Since,  $V_{w_2} < u$ , the outlet velocity triangle is modified as follows:



$$(i) F_x = \rho \cdot a \cdot V_1 [V_{w_1} - V_{w_2}]$$

$$\boxed{F_x = 104.2 \text{ kN}}$$

$$(ii) \text{ Power developed, } P = \frac{F_x \times u}{1000}$$

$$\boxed{P = 4247.15 \text{ kW}}$$



1.(c)		5	K2	CO2
Sol	Turbomachine - Definition	-1-		
	- Neat sketch	-1-		
	- Explanation of basic parts			
	↳ Rotor ↳ Stator ↳ Casing ↳ Input or output shaft	-3-		

OR

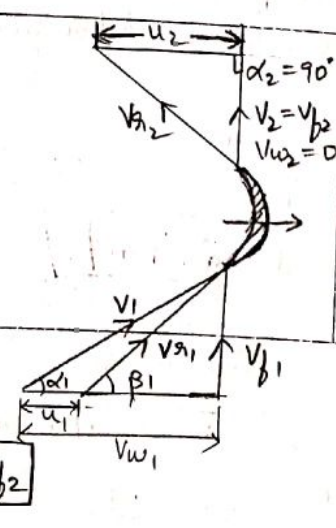
2(a)		5	K2	CO1
Sol	Definitions of all efficiencies	-3-		
	Explanation of all the terms along with units	-2-		

2.(b)		5	K3	CO1
Sol	<p>Given:</p> <p>No. of jets = 2</p> <p>Power generated = 7500 kW</p> <p>Net Head, H = 400m</p> <p><math>\eta_o = 0.8</math></p> <p><math>\phi = 0.47</math></p> <p>(i) diameter of each jet, d = ?</p> <p>(ii) Total flow rate, <math>Q_{total} = ?</math></p> <p>(iii) Force exerted in tangential direction, <math>F_x = ?</math></p> <p>W.K.T <math>\eta_o = \frac{P}{\frac{\rho g Q_{Total} H}{1000}}</math></p> <p><math>\Rightarrow Q_{Total} = 2.41 \text{ m}^3/\text{s}</math></p>	-1-		
	<p>Angle of deflection = <math>(180 - \beta_2) = 165^\circ</math></p> <p><math>\Rightarrow \beta_2 = 15^\circ</math></p> <p><math>V_{r2} = (1 - 0.15) V_{r1}</math></p> <p><math>V_{r2} = 0.85 V_{r1}</math></p> <p><math>C_v = 0.97</math></p>	-1-		

$V_{w1} = V_1 = C_v \sqrt{2gH} = 85.93 \text{ m/s}$   
 $u_1 = u_2 = u = \phi \sqrt{2gH} = 41.63 \text{ m/s}$   
 $V_{r1} = V_1 - u = 44.3 \text{ m/s}$   
 $V_{r2} = 0.85 (44.3) = 37.65 \text{ m/s}$

	<p>(i) Discharge through 1 jet = <math>\frac{Q_{Total}}{\text{no. of jets}} = \frac{1.2 \text{ m}^3/\text{s}}{9} \Rightarrow d = 0.133 \text{ m}</math></p> <p>(ii) <math>F_x = \rho a v_1 [v_{w1} + v_{w2}]</math>  <math>= \rho a v_1 [v_{w1} - v_{w2}]</math></p> <p><math>F_x = 96.376 \text{ kN}</math></p> <p>From outlet velocity triangle,  <math>v_{w2} = v_{r2} \cos \beta_2 - u</math>  <math>v_{w2} = -5.2 \text{ m/s}</math></p> <p>Since <math>v_{w2} &lt; u_2</math>, modified velocity triangle is drawn.</p>	-1-		
2(c)		5	K2	CO2
Sol	Detailed classification of turbomachines	-5-		

PART-B

3(a)		5	K2	CO1
Sol	<p>→ Neat sketch of Francis Turbine</p> <p>→ Explanation for basic parts of Francis Turbine</p> <p>→ Inlet and outlet velocity triangles with proper representation of velocities.</p>	-1- -2- -2-		
3.(b)		5	K3	CO1
Sol	<p>Given: <math>H = 70 \text{ m}</math>      <math>\eta_H = 0.95</math>  <math>N = 600 \text{ rpm}</math>      <math>k_b = 0.25</math>  <math>P = 368 \text{ kW}</math>  <math>\eta_o = 0.85</math>      <math>\frac{B_1}{D_1} = 0.1</math>  <math>D_1 = 2 \times D_2</math>      <math>v_{f1} = v_{f2}</math>          Area blocked = 10%</p> <p>WKT <math>\eta_o = \frac{P}{\frac{\rho g Q H}{1000}} \Rightarrow Q = 0.6304 \text{ m}^3/\text{s}</math></p> <p><math>k_b = \frac{v_{f1}}{\sqrt{2gH}} \Rightarrow v_{f1} = 9.264 \text{ m/s} = v_{f2}</math></p> 	-1-		



Also,  $Q = \pi D_1 B_1 \times v_{b1}$

$Q = A_f \times v_{b1}$

w.k.T  $A_f = (1 - 0.1)$  because, 10% is blocked.

$\therefore Q = 0.9 A_f \times v_{b1}$

$\Rightarrow A_f = 0.0756 \text{ m}^2 \Rightarrow D_1 = 0.4905 \text{ m} \Rightarrow B_1 = 0.049 \text{ m}$

$\therefore D_f = 2 \times D_2 \Rightarrow D_2 = 0.245 \text{ m}$

From inlet velocity triangle,

$\tan \alpha_1 = \frac{v_{b1}}{v_{w1}} \Rightarrow \alpha_1 = 12.33^\circ$

$\tan \beta_1 = \frac{v_{b1}}{(v_{w1} - u_1)} \Rightarrow \beta_1 = 18.96^\circ$

From outlet velocity triangle,

$\tan \beta_2 = \frac{v_{b2}}{u_2}$

$\Rightarrow \beta_2 = 50.304^\circ$

$u_1 = \frac{\pi D_1 N}{60} = 15.4 \text{ m/s}$

$u_2 = \frac{\pi D_2 N}{60} = 7.69 \text{ m/s}$

$z_H = \frac{v_{w1} u_1}{gH}$

$\Rightarrow v_{w1} = 42.36 \text{ m/s}$

(i)  $\alpha_1 = 12.33$ , (ii)  $\beta_1 = 18.96^\circ$ ,  $\beta_2 = 50.304^\circ$ , (iii)  $D_2 = 0.245 \text{ m}$  (iv)  $B_1 = 0.049 \text{ m}$

3.(c)

5

k<sub>2</sub>

CO<sub>2</sub>

Differences between a turbomachine and a positive displacement machine:

→ Differences with respect to different operating conditions along with examples.

Sol

-5-

4(a)

- Sol → Neat sketch of Kaplan Turbine  
 → Explanation of the functions of each part  
 → Drawing the velocity triangles for Kaplan turbine

-1-

-2-

-2-

4(b)

Sol

5

K3

CO1

Given: Head,  $H = 20\text{ m}$ Power,  $P = 11775\text{ kW}$ Outer dia,  $D_o = 3.5\text{ m}$ Hub dia,  $D_b = 1.75\text{ m}$ Guide blade angle,  $\alpha_1 = 35^\circ$  $\gamma_H = 0.88$  $\gamma_o = 0.84$  $V_{w2} = 0$ 

$$\text{W.K.T } \gamma_o = \frac{P}{\frac{\rho g Q H}{1000}} \Rightarrow \boxed{Q = 71.44\text{ m}^3/\text{s}}$$

$$\text{Also, } Q = \frac{\pi}{4} [D_o^2 - D_b^2] \times V_{f1}$$

$$\Rightarrow \boxed{V_{f1} = 9.9\text{ m/s}} = V_{f2}$$

From inlet velocity triangle,

$$\tan \alpha_1 = \frac{V_{f1}}{V_{w1}} \Rightarrow \boxed{V_{w1} = 14.14\text{ m/s}}$$

$$\text{W.K.T } \gamma_H = \frac{V_{w1} u_1}{g H} \Rightarrow \boxed{u_1 = 12.20\text{ m/s}} = u_2$$

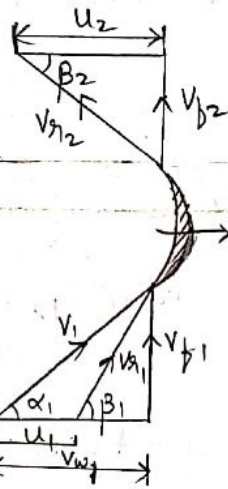
$$\text{(i) Now, } \tan \beta_1 = \frac{V_{f1}}{(V_{w1} - u_1)} \Rightarrow \boxed{\beta_1 = 78.96^\circ}$$

From outlet velocity triangle,

$$\text{(ii) } \tan \beta_2 = \frac{V_{f2}}{u_2} \Rightarrow \boxed{\beta_2 = 39.05^\circ}$$

$$\text{(iii) Speed of turbine } \Rightarrow u_1 = u_2 = u = \frac{\pi D_o N}{60}$$

$$\Rightarrow \boxed{N = 66.57\text{ rpm}}$$





A(c)

Sol

Power Generating	Power Absorbing Turbomachine	5	K2	CO2
(i) Steam Turbine (ii) Wind turbine	(i) Fan (ii) Air blower (iii) Axial flow compressor.	-5-		

*K. Rama*  
Course In charge

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